

## Pyromet<sup>®</sup> Alloy 718

## Unit Display: English



E-Mail Datasheet Add to My Materials

	Identification	
UNS Number		
<ul> <li>N07718</li> </ul>		
DIN Number		
• 0.4440		

2.4668

Type Analysis							
Carbon	0.08 %	Manganese	0.35 %				
Phosphorus	0.015 %	Sulfur	0.015 %				
Silicon	0.35 %	Chromium	17.00 to 21.00 %				
Molybdenum	2.80 to 3.30 %	Copper	0.15 %				
Cobalt	1.00 %	Titanium	0.65 to 1.15 %				
Aluminum	0.35 to 0.80 %	Columbium + Tantalum	4.75 to 5.50 %				
Boron	0.001 to 0.006 %	Nickel + Cobalt	50.00 to 55.00 %				
Iron	11.30 to 22.50 %						

The Cobalt content is on an "If determined" basis.

## **General Information**

#### Description

Pyromet<sup>®</sup> alloy 718 is a precipitation hardenable nickel-base alloy designed to display exceptionally high yield, tensile and creep-rupture properties at temperatures up to 1300°F (704°C). The sluggish age-hardening response of Pyromet alloy 718 permits annealing and welding without spontaneous hardening during heating and cooling. This alloy has excellent weldability when compared to the nickel-base superalloys hardened by aluminum and titanium.

## Applications

This alloy has been used for jet engine and high-speed airframe parts such as wheels, buckets, spacers, and high temperature bolts and fasteners.

#### **Corrosion Resistance**

**Important Note:** The following 4-level rating scale is intended for comparative purposes only. Corrosion testing is recommended; factors which affect corrosion resistance include temperature, concentration, pH, impurities, aeration, velocity, crevices, deposits, metallurgical condition, stress, surface finish and dissimilar metal contact.

Nitric Acid	Good	Sulfuric Acid	Moderate
Phosphoric Acid	Moderate	Acetic Acid	Good
Sodium Hydroxide	Good	Salt Spray (NaCl)	Excellent
Sea Water	Moderate	Sour Oil/Gas	Good
Humidity	Excellent		

0.2970 lb/in <sup>3</sup>
0.2070 lb/in3
0.2070 lb/in3
0.2970 10/113
7.10 x 10 <sup>-6</sup> in/in/°F
7.50 x 10 <sup>-6</sup> in/in/°F
7.70 x 10 <sup>-6</sup> in/in/°F
7.90 x 10 <sup>-6</sup> in/in/°F
8.00 x 10 <sup>-6</sup> in/in/°F
8.40 x 10 <sup>-6</sup> in/in/°F
8.90 x 10 <sup>-6</sup> in/in/°F

# **Coefficient of Thermal Expansion**

Temperatu	ure Range	10 <sup>.6</sup> /°F	10-5/°C
77°F to	25°C to	1077	10-7-0
200	93	7.1	12.8
400	204	7.5	13.5
600	316	7.7	13.9
800	427	7.9	14.2
1000	538	8.0	14.4
1200	649	8.4	15.1
1400	760	8.9	16.0

## Modulus of Elasticity (E)

would of Elasticity (E)	
70°F	29.6 x 10 <sup>3</sup> ksi
200°F	29.2 x 10 <sup>3</sup> ksi
400°F	28.8 x 10 <sup>3</sup> ksi
600°F	27.6 x 10 <sup>3</sup> ksi
800°F	26.5 x 10 <sup>3</sup> ksi
1000°F	25.5 x 10 <sup>3</sup> ksi
1200°F	24.5 x 10 <sup>3</sup> ksi
1400°F	23.1 x 10 <sup>3</sup> ksi
1600°F	18.1 x 10 <sup>3</sup> ksi
1750°F	11.1 x 10 <sup>3</sup> ksi

# Modulus of Elasticity (E)

Tempe	rature		MD= + 101
°F	°C	psix 10⁵	MPa x 10 <sup>3</sup>
70	21	29.6	2.08
200	93	29.2	2.05
400	204	28.8	2.02
600	316	27.6	1.94
800	427	26.5	1.86
1000	538	25.5	1.79
1200	649	24.5	1.72
1400	760	23.1	1.62
1600	871	18.1	1.27
1750	954	11.1	0.78

Electrical Resistivity	
70.0°F	728.0 ohm-cir-mil/ft
Melting Range	
	2200.000 to 2450.000 °F

**Typical Mechanical Properties** 

Effect of Solution Treating Temperature on Transverse Room Temperature Properties – Pyromet Alloy 718 Tensile properties of a 4" (101.6 mm) square billet

Solution Treating Temperature		0.2% Yield Strength		Tensile Strength		% Elongation	% Reduction	Rockwell C	ASTM Grain
۴F	°C	ksi	MPa	ksi	MPa	in 2" (50.8 mm)	of Area	Hardness	Size
1750	954	160.1	1104	183.7	1267	8.1	13.6	42	7
1900	1038	163.7	1129	183.7	1267	12.8	20.3	42	5/6
1950	1066	158.2	1091	180.9	1247	14.0	19.6	42	4/5
2000 2050	1093 1121	154.6 149.9	1066 1033	178.0 171.1		17.3 17.5	24.5 27.9	40 39	4/4.5 3.5/4

Note: All samples solution treated 2 hours at the temperature indicated and air cooled. Aging consisted of 8 hours at 1325°F (718°C), cooled 100°F/hr to 1150°F (56°C/hr to 621°C), held 8 hours at 1150°F (621°C), and air cooled.

## Effect of Solution Treating Temperature on the Transverse Stress Rupture Properties – Pyromet Alloy 718

4" (101.6 mm) square billet

Solution Treating Temperature		Test Temperature		Stress		Rupture Life	% Elongation	% Reduction
۴F	°C	٩F	°C	ksi	MPa	Hours	in 2" (50.8 mm)	of Area
1750 1900 1950 2000	954 1038 1066 1093	1300 1300 1300 1300	704 704 704 704	75 75 75 75	517 517 517 517 517	64.8 132.1 95.2 141.0	17.4 9.4 7.9 2.5	25.3 17.2 10.8 6.8

Notes: All samples solution treated 2 hours at the temperatures indicated and air cooled. Aging consisted of 8 hours at 1325°F (718°C), cooled 100°F/hr to 1150°F (56°C/hr to 621°C), held 8 hours at 1150°F (621°C), and air cooled.

All specimens were combination smooth-notch bars with 0.178" (4.52 mm) gage diameter, 0.712" (18.1 mm) gage length and  $K_t = 3.8$ .

Stress rupture notch sensitivity may result from 1900°F (1038°C) solution treatment. This is due to reduction of precipitate at grain boundaries with this treatment.

Elevated Temperature Tensile Tests – Pyromet Alloy 718 1/2" (12.7 mm) bar stock

	Test Temperature		0.2% Yield Strength		sile ngth	% Elongation	
°F	°C	ksi	MPa	ksi	MPa	2" (50.8 mm)	
70	21	175	1207	210	1448	22.0	
200	93	170	1172	204	1407	21.0	
400	204	163	1124	198	1365	20.0	
600	316	159	1096	195	1344	20.0	
800	427	156	1076	191	1317	19.0	
1000	538	155	1069	185	1276	18.0	
1200	649	149	1027	168	1158	19.0	
1400	760	110	758	110	758	27.0	

Heat Treatment: 1 hr—1800°F (982°C), air cool + 8 hr—1325°F (718°C), cool 100°F/hr to 1150°F (56°C/hr to 621°C), hold 8 hr, air cool.

## Low Temperature Tensile Data – Pyromet Alloy 718 Transverse tests—4" (101.6 mm) square billet

Test No.		est erature		0.2% /ield Strength		sile ngth	% Elongation	% Reduction
	۴F	°C	ksi	MPa	ksi	MPa	in 2" (50.8 mm)	of Area
1 2 3 4 5 6	Room Room -423 -423 -423	Room Room -253 -253 -253	157 156 152.5 196.2 200 194.2	1082 1076 1051 1353 1379 1339	183.8 181.3 179.5 250.4 244.2 241.9	1267 1250 1238 1726 1684 1668	18.0 16.5 19.0 14.0 13.0 12.0	21.6 23.3 29.1 16.2 17.5 16.7

Heat Treatment: 2 hr—1950°F (1066°C), air cool + 8 hr—1325°F (718°C), cool 100°F/hr to 1150°F (56°C/hr to 621°C), hold 8 hr, air cool.

## Stress Rupture Data - Pyromet Alloy 718

				Stress for Rupture						
Test Temperature		100 Hours				1000 Hours				
		Sme	ooth	Notch		Smo	nooth Notch		tch	
°F	°C	ksi	MPa	ksi	MPa	ksi	MPa	ksi	MPa	
1100 1200 1300 1400	593 649 704 760	170 110 75 44	1172 758 517 303	220 195 130 63	1517 1344 896 434	130 85 55 25	896 586 379 172	205 170 80 35	1413 1172 552 241	

Heat Treatment: 1 hr—1800°F (982°C), air cool + 8 hr—1325°F (718°C), furnace cool 100°F/hr to 1150°F (56°C/hr to 621°C), hold 8 hr, air cool.

### **Heat Treatment**

The following heat treatment should be used to obtain the best combination of tensile properties and stress rupture properties:

1 hr 1750°F to 1800°F (954°C to 982°C) air cool + 8 hr 1325°F (718°C) cool 100°F/hr to 1150°F (56°C/hr to 621°C), hold 8 hr and air cool.

To obtain the best room temperature and cryogenic tensile properties, the following heat treatment should be used:

1 to 2 hr 1950°F (1066°C), air cool + 8 hr 1325°F (718°C) cool 100°F/hr to 1150°F (56°C/hr to 621 °C), hold 8 hr and air cool.

## Workability

#### Hot Working

Hot working is carried out using a 2050°F (1121°C) maximum furnace temperature. Hot-cold working in the range 1700/1850°F (927/1010°C) will improve the strength of the forging if the service temperature is below about 1100°F (593°C). Prolonged soaking at the forging temperature is not desirable. The material should be given uniform reductions to avoid the formation of duplex grain structures.

#### Machinability

The alloy can be readily machined in either the annealed or the age-hardened condition. The age-hardened condition gives better chip action on chip breaker tools and produces a better finish. The annealed condition will give a slightly longer tool life. Tooling and procedures are similar to those used for Carpenter Pyromet alloy X-750.

#### Weldability

Pyromet alloy 718 can be welded in either the annealed or the aged condition. Welding in the aged condition will cause the formation of a softened heat-affected zone.

Other Information							
Applicable Specifications							
• AMS 5662	• AMS 5663						
Forms Manufactured							
Bar-Rounds	Bar-Shapes						
Billet	Strip						

http://cartech.ides.com/datasheet.aspx?i=101&c=TechArt&E=141&FMT=PRINT

• Wire

- Technical Articles
  - A Designer's Manual On Specialty Alloys For Critical Automotive Components
     A Cuide to Etching Specialty Alloys for Microstructural Evolution
  - A Guide to Etching Specialty Alloys for Microstructural Evaluation
     Alloy Selection for Cold Forming (Part I)
  - Alloy Selection for Cold Forming (Part I)
     Alloy Selection for Cold Forming (Part II)
  - Forging Difficult Alloys: How to Get Better Results, Consistently
  - How to Select the Right Stainless Steel or High Temperature Alloy for Heading
  - New Requirements for Ferrous-Base Aerospace Alloys
  - Selecting High Temperature Alloys for Fasteners in Automotive Exhaust Systems
  - Trends in High Temperature Alloys

### Disclaimer:

The information and data presented herein are typical or average values and are not a guarantee of maximum or minimum values. Applications specifically suggested for material described herein are made solely for the purpose of illustration to enable the reader to make his/her own evaluation and are not intended as warranties, either express or implied, of fitness for these or other purposes. There is no representation that the recipient of this literature will receive updated editions as they become available.

Unless otherwise specified, registered trademarks are property of CRS Holdings Inc., a subsidiary of Carpenter Technology Corporation. Copyright 2009 CRS Holdings Inc. All rights reserved.

Edition Date: 10/16/08